



**Jet Propulsion Laboratory**  
California Institute of Technology

# Exploring the sensitivity of satellite gravimetry to temporal aliasing errors

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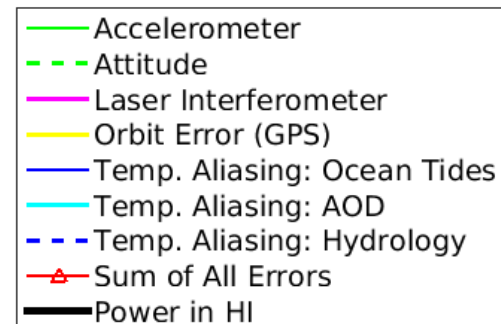
<sup>1</sup>Jet Propulsion Laboratory, California Institute of Technology

October 10, 2017  
GRACE Science Team Meeting  
Austin, Texas

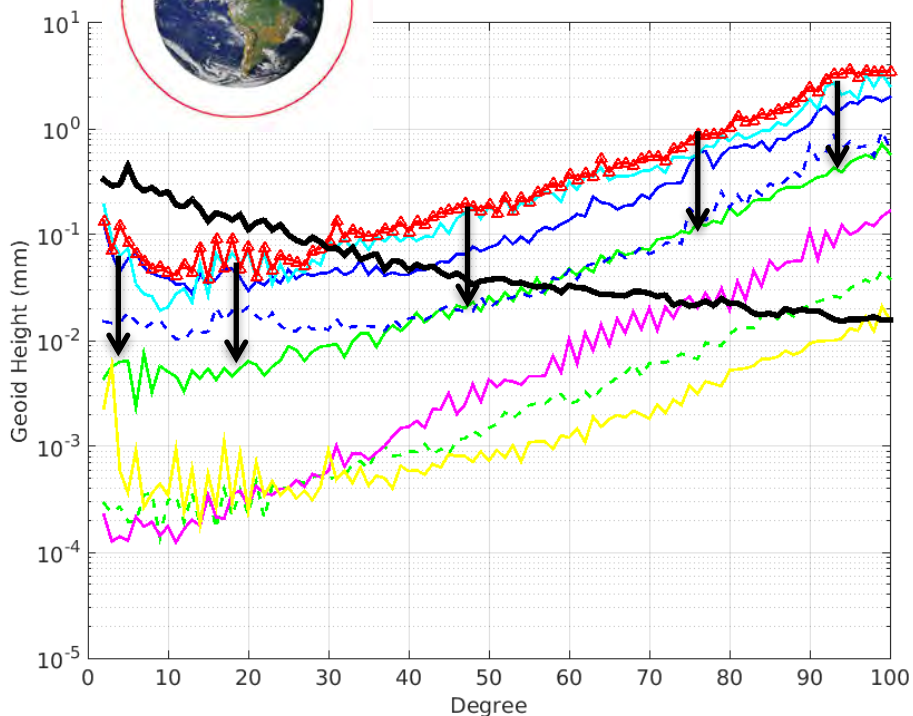
Simulation Results: **500 km altitude**, 300 km separation, GRACE-FO measurement errors

## Temporal aliasing errors are dominant for single and dual-pair architectures

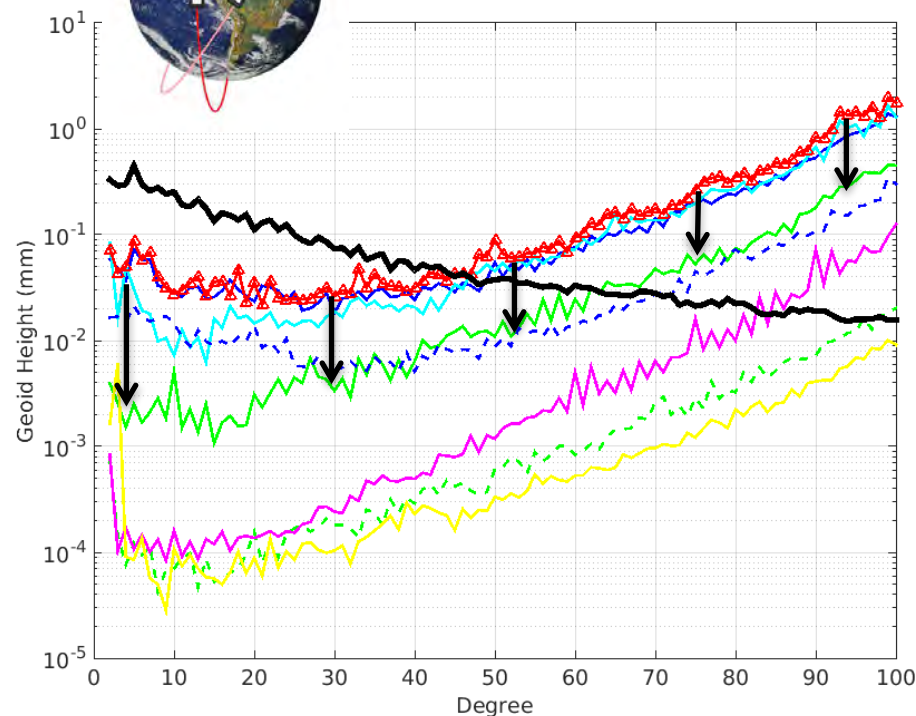
*Can we reduce aliasing error to here?*



One Pair

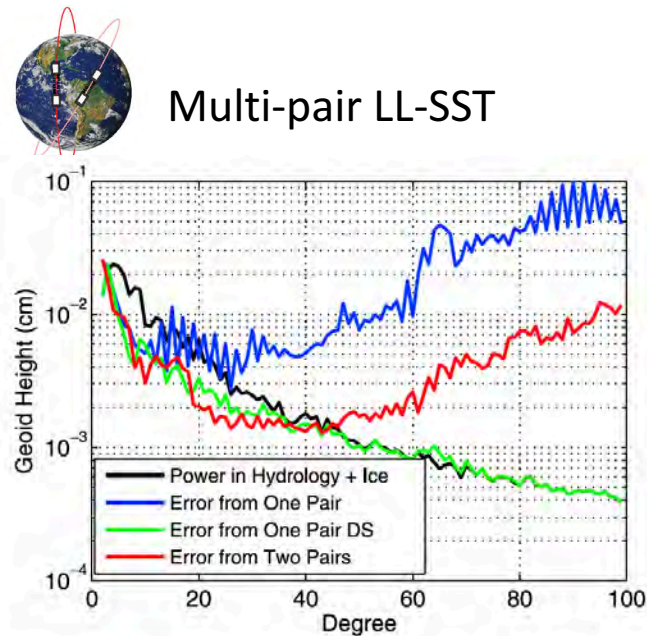


Two Pairs



# How can we reduce temporal aliasing errors?

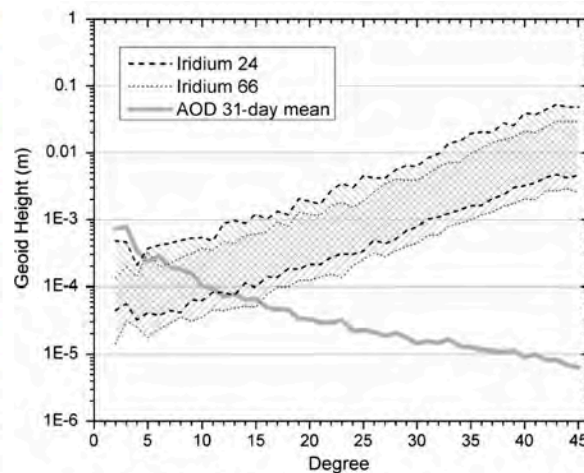
- 1) Improve background force models
- 2) Co-estimate parameters
- 3) Sample the gravity field more frequently**



**Figure 4.** Geoid degree error for architectures consisting of one and two pairs of satellites.

*Wiese et al., 2011b*

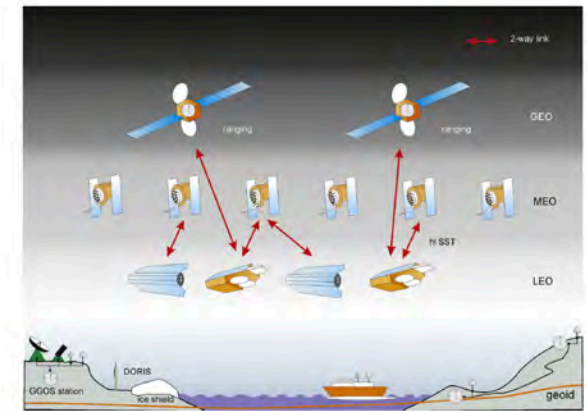
Constellation of GNSS receivers



**Fig. 6** Degree variances of the total error bands for the 31-day Iridium 14- and 66-satellite constellation simulations using assumptions for high and low measurement noise.

*Gunter et al., 2011*

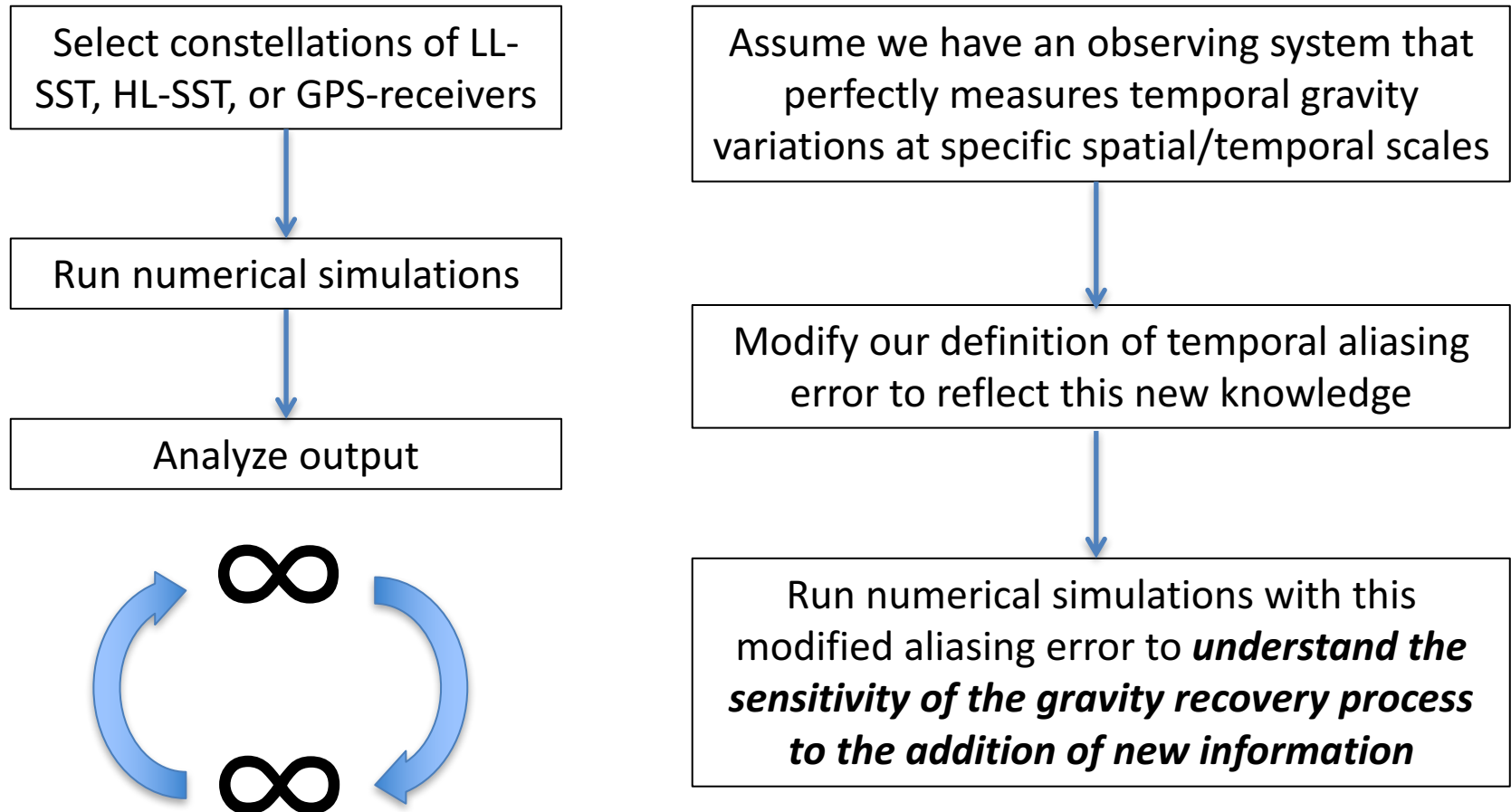
GETRIS: HL-SST Concept



GETRIS constellation for gravity field recovery. The advantages can be seen in the LEO orbit which makes every mission the data link is also a high precision ranging link. This allows a continuous gravity field recovery with high temporal resolution.

*Hauk et al., 2016*

# Goal: Identify an observing system for which temporal aliasing errors are not the limiting source of error in recovering time variable gravity



# Simulation Setup

	Truth Model	Nominal Model
Static Gravity Field	gif48	gif48
Ocean Tides	GOT4.7	FES04
Atmosphere/Ocean	"AO"	AOerr+DEAL
Hydrology/Ice	"HI"	--

AOHIS 3-hr temporal res. *Dobslaw et al.* 2015

- 29-day simulation
  - January 2006
- Mission Architectures
  - Single Polar Pair
  - Polar Pair + Lower Inclined Pair (72°)
- Altitudes
  - ~ 300, 500 km
- Separation Distance:
  - 300 km
- All satellites are in exact 29-day repeat orbits
- Simulation carried out to  $n = 100$

## Instrument noise models

- Laser (LRI) GRACE-FO Requirement
  - RMS of 10.88 nm/s
- Accelerometer GRACE-FO CBE

$$\text{Alongtrack, Radial: } 3.21 \times 10^{-11} \sqrt{1 + \frac{.01}{f} + 20f^4} \text{ m/s}^2/\text{Hz}^{1/2}$$

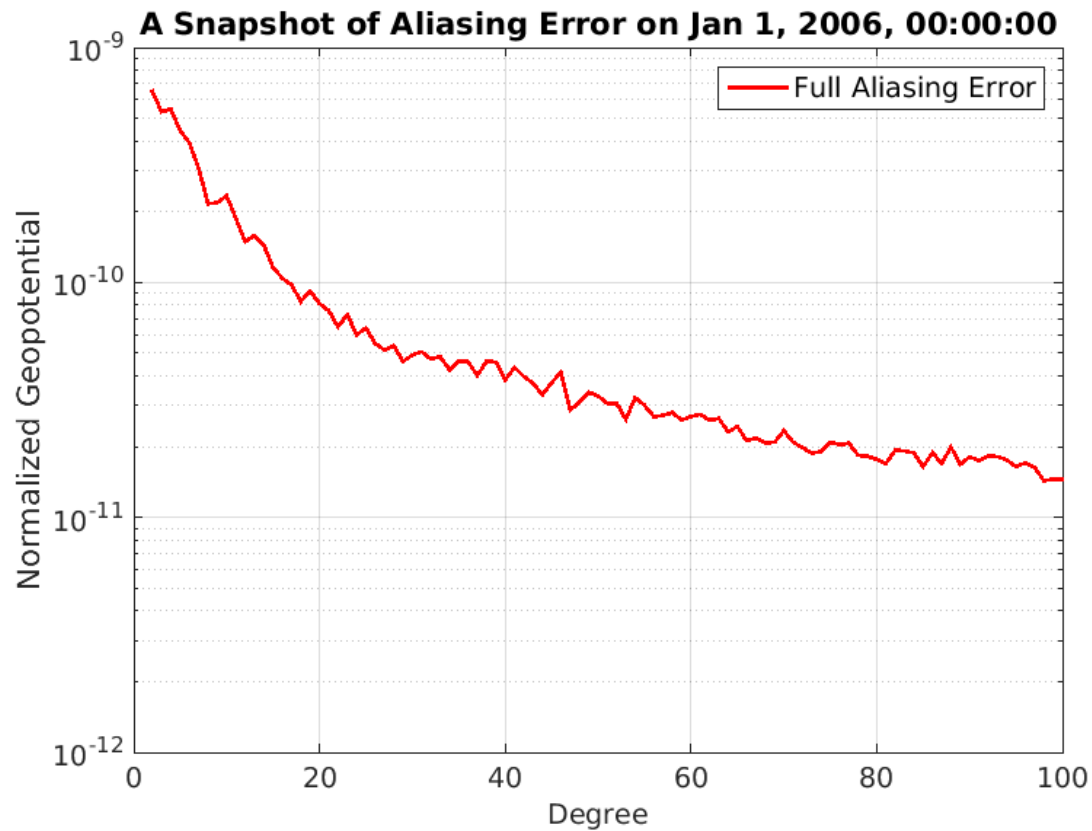
$$\text{Crosstrack: } 4.75 \times 10^{-10} \sqrt{1 + \frac{.01}{f} + 5 \times 10^{-4} f^4} \text{ m/s}^2/\text{Hz}^{1/2}$$

- Attitude GRACE-FO Requirement
  - Impacts ONLY accelerometer error, not MWI or LRI pointing

$$\text{Pitch, Roll: } 2.1 \times 10^{-5} \text{ rad/Hz}^{1/2}$$

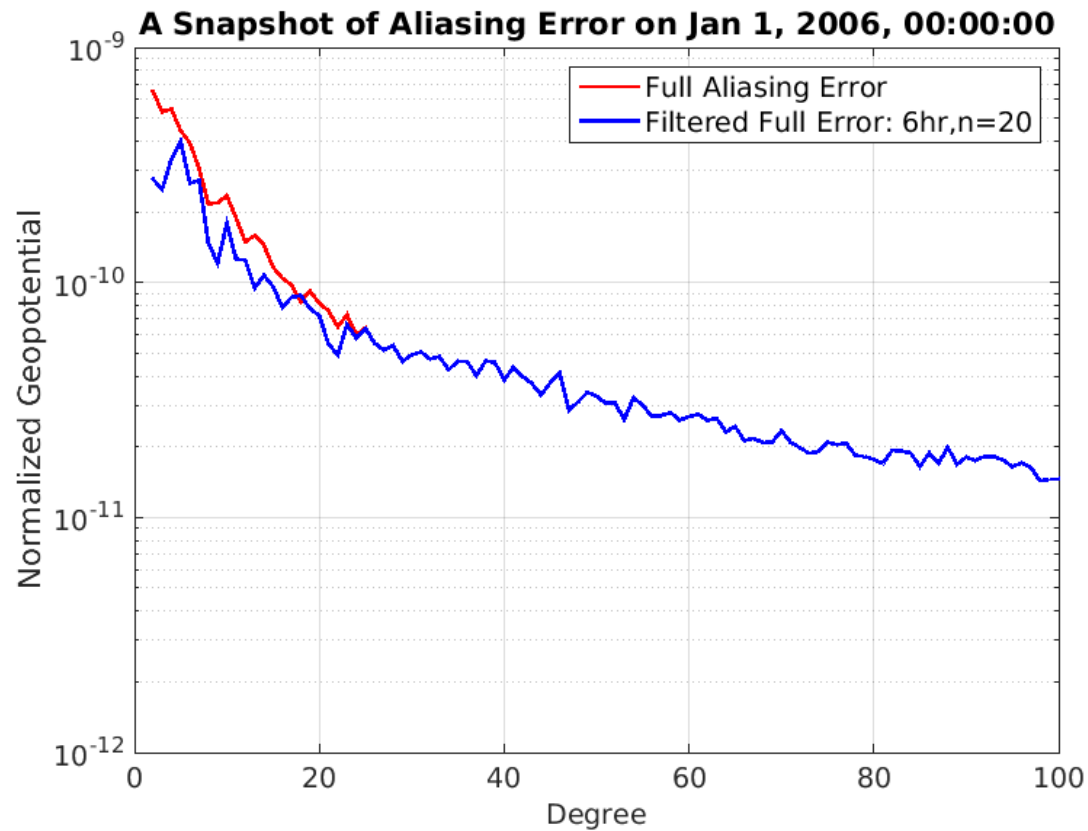
$$\text{Yaw: } 1.7 \times 10^{-4} \text{ rad/Hz}^{1/2}$$

- GPS data is mimicked by using satellite positions as observables
  - 1 cm white noise in 3-axes is added



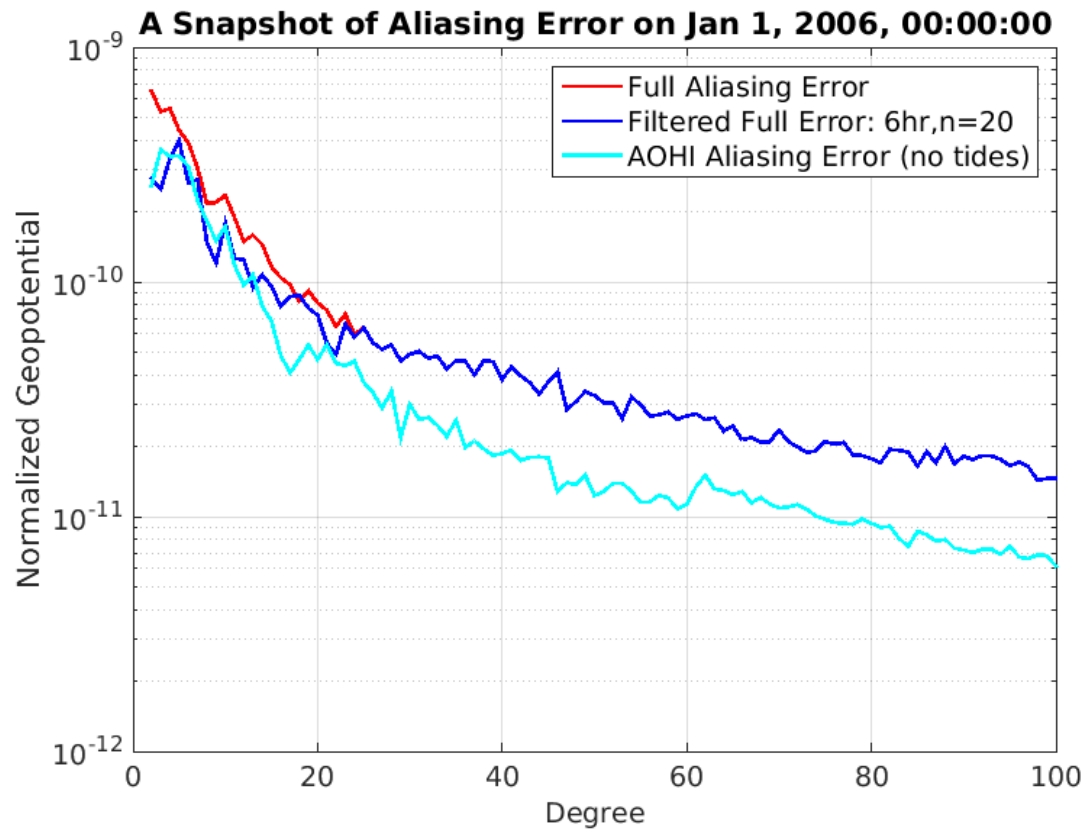
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<b>Atmosphere/Ocean</b>	"AO"	AOerr+DEAL
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AOHIS 3-hr temporal res. *Dobslaw et al. 2015*



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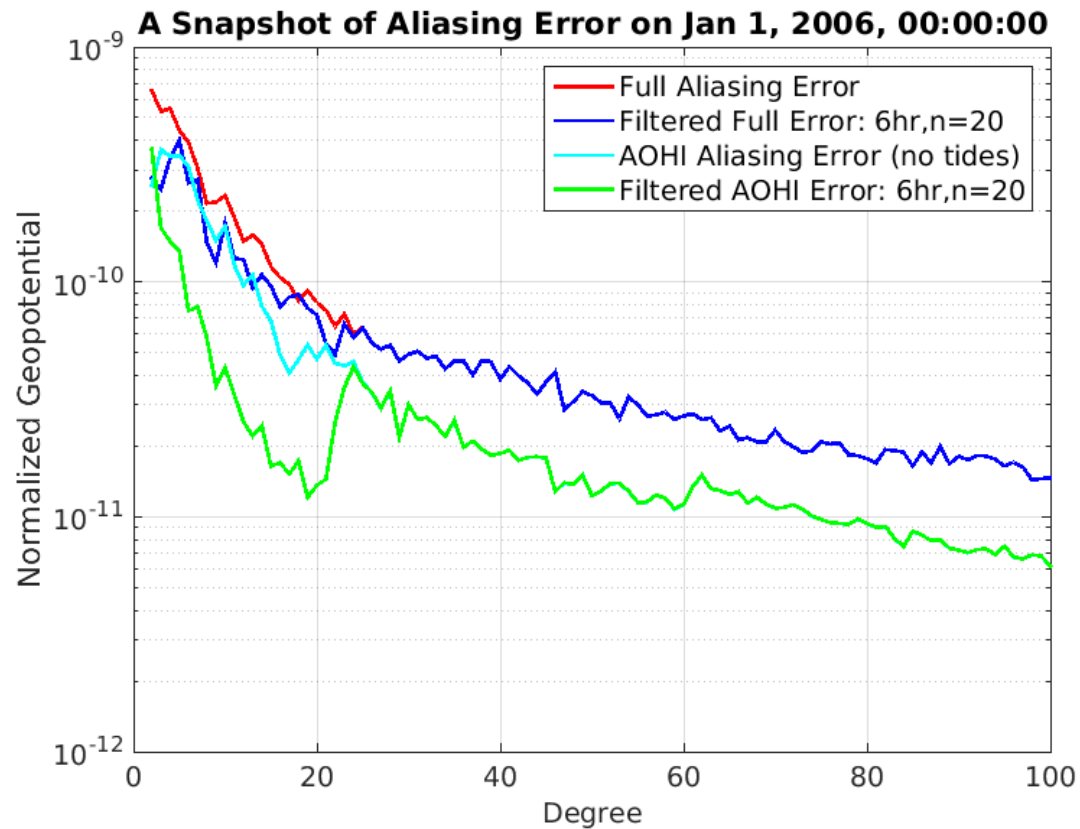
AOHIS 3-hr temporal res. *Dobslaw et al. 2015*



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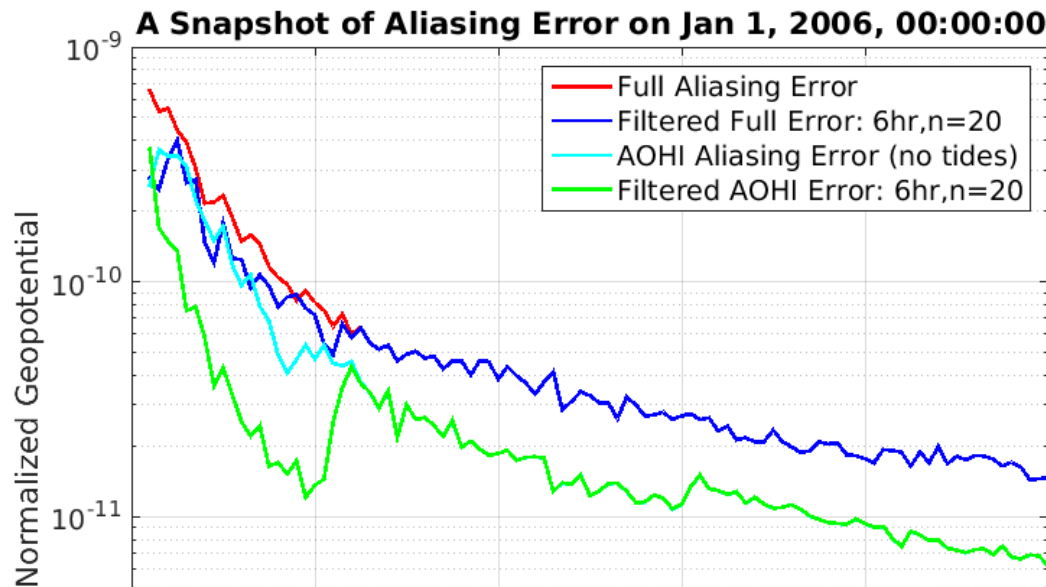
AOHIS 3-hr temporal res. *Dobslaw et al. 2015*





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AOHIS 3-hr temporal res. *Dobslaw et al.* 2015



We assume that we have perfect knowledge of mass variations at time scales of **[6, 12, 24] hours** and at spatial scales of **n = [10, 20, 30]**.

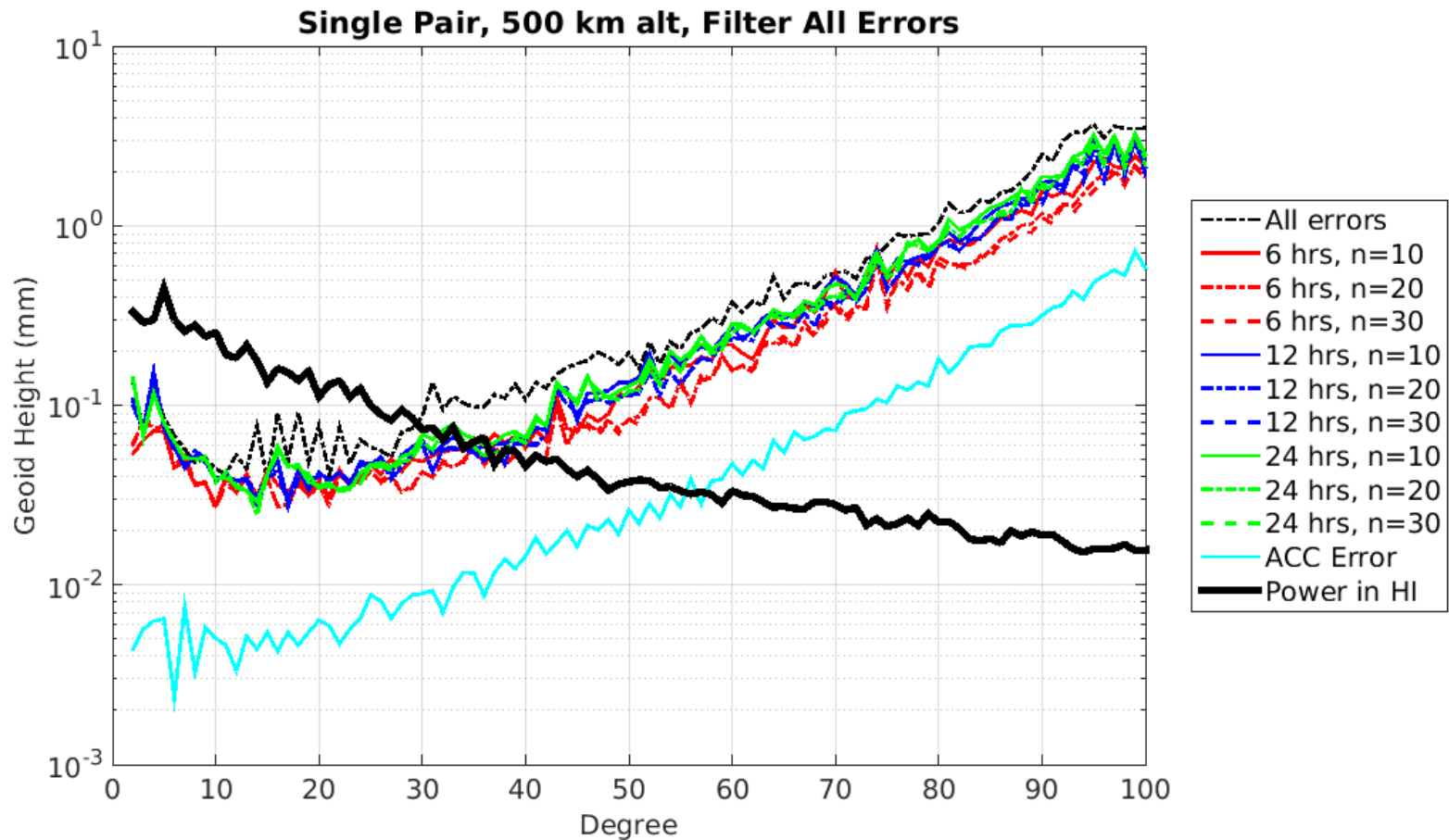
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AOHIS 3-hr temporal res. *Dobslaw et al.* 2015

Architecture: One Pair

Altitude: 500 km

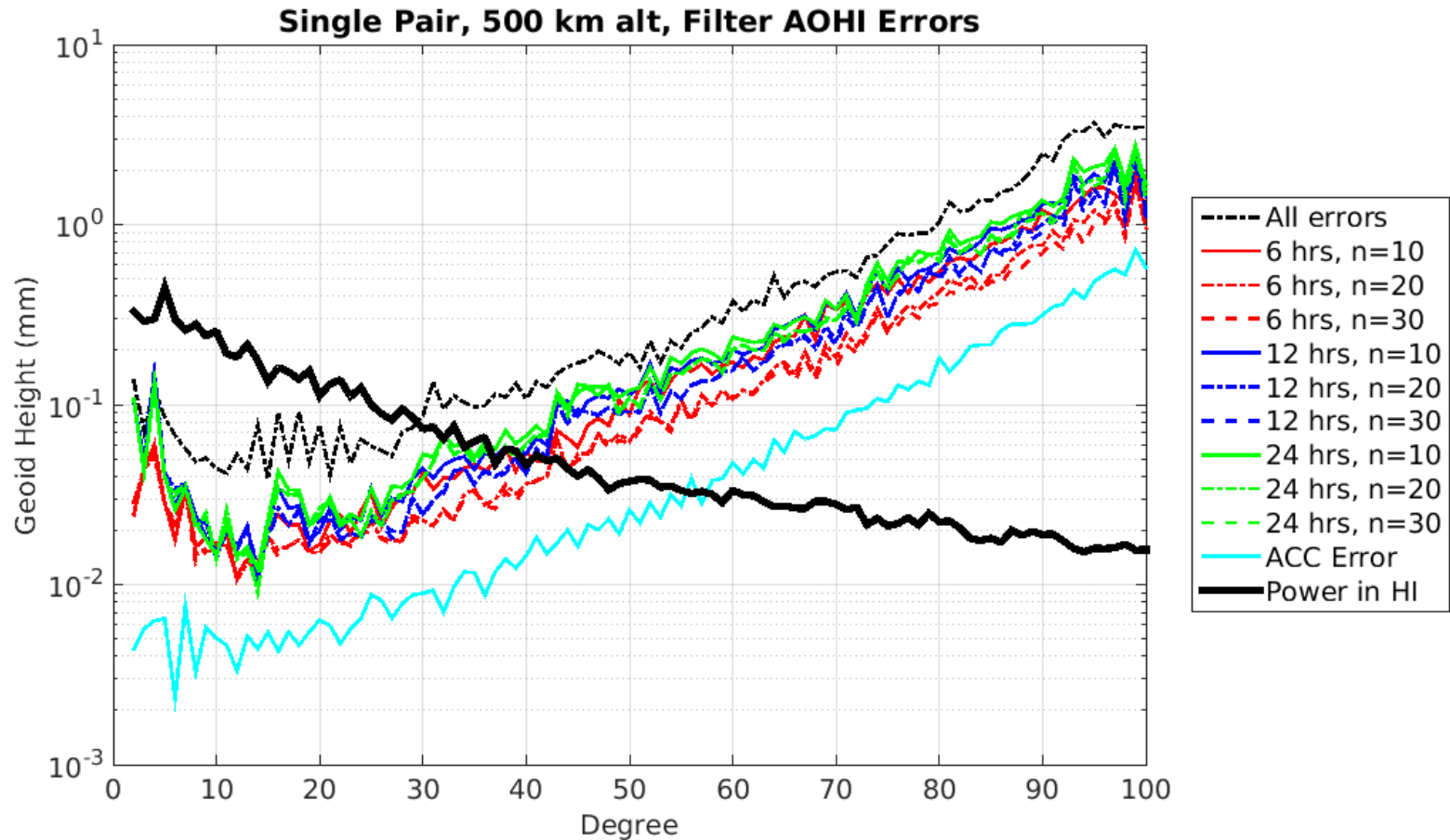
Aliasing: OT + AOHI



Architecture: One Pair

Altitude: 500 km

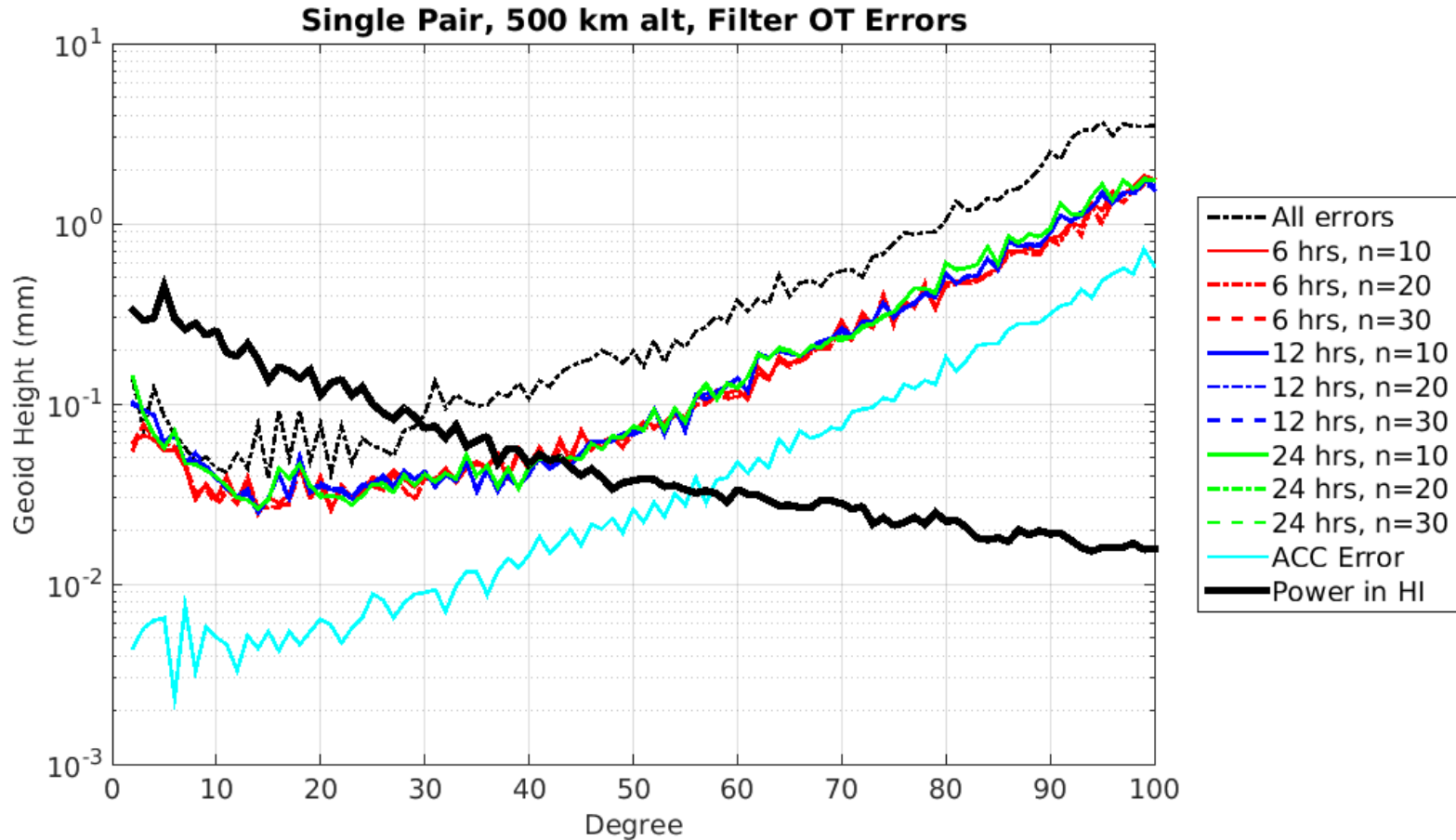
Aliasing: AOHI



Architecture: One Pair

Altitude: 500 km

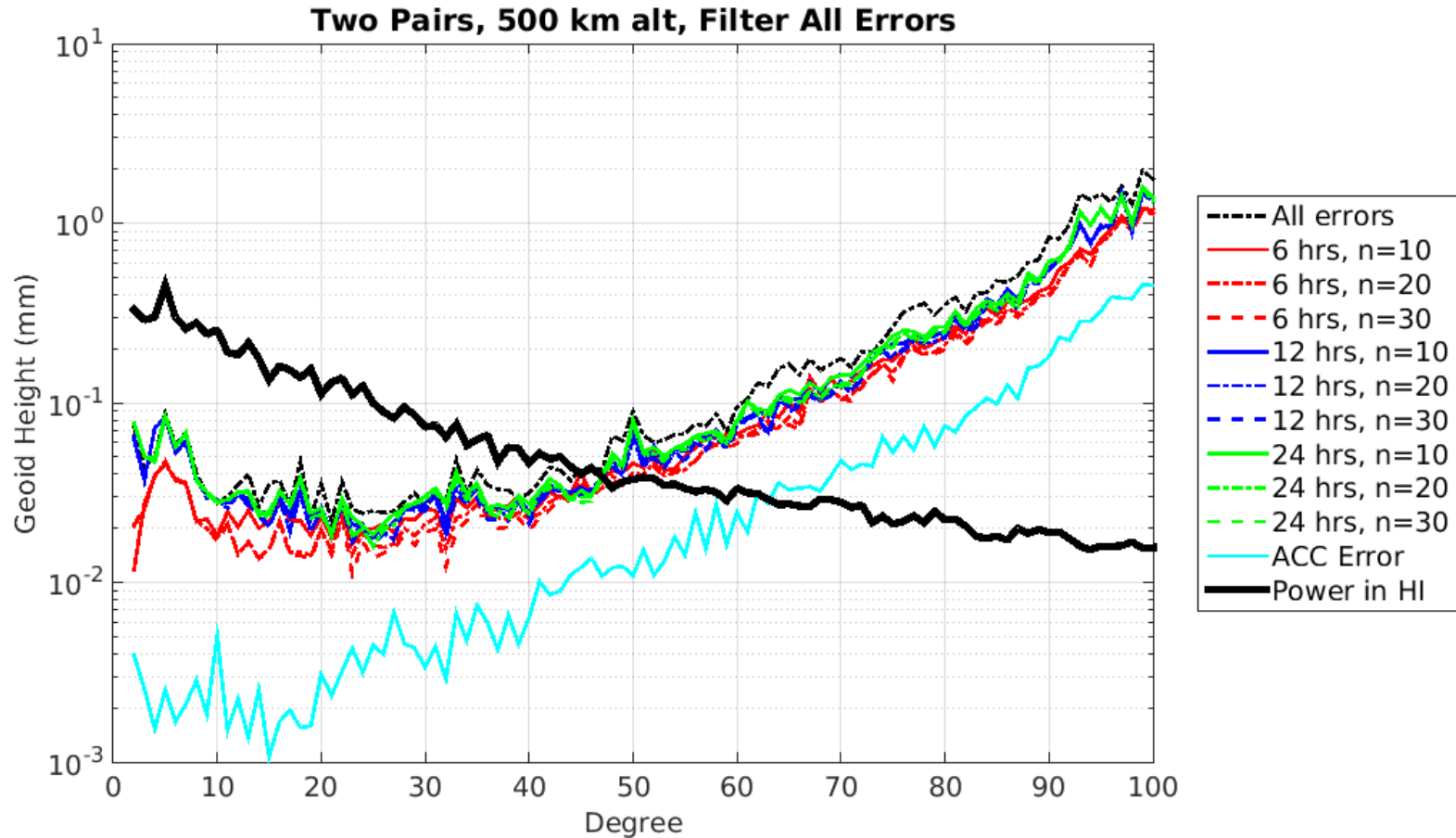
Aliasing: OT



Architecture: Two Pairs

Altitude: 500 km

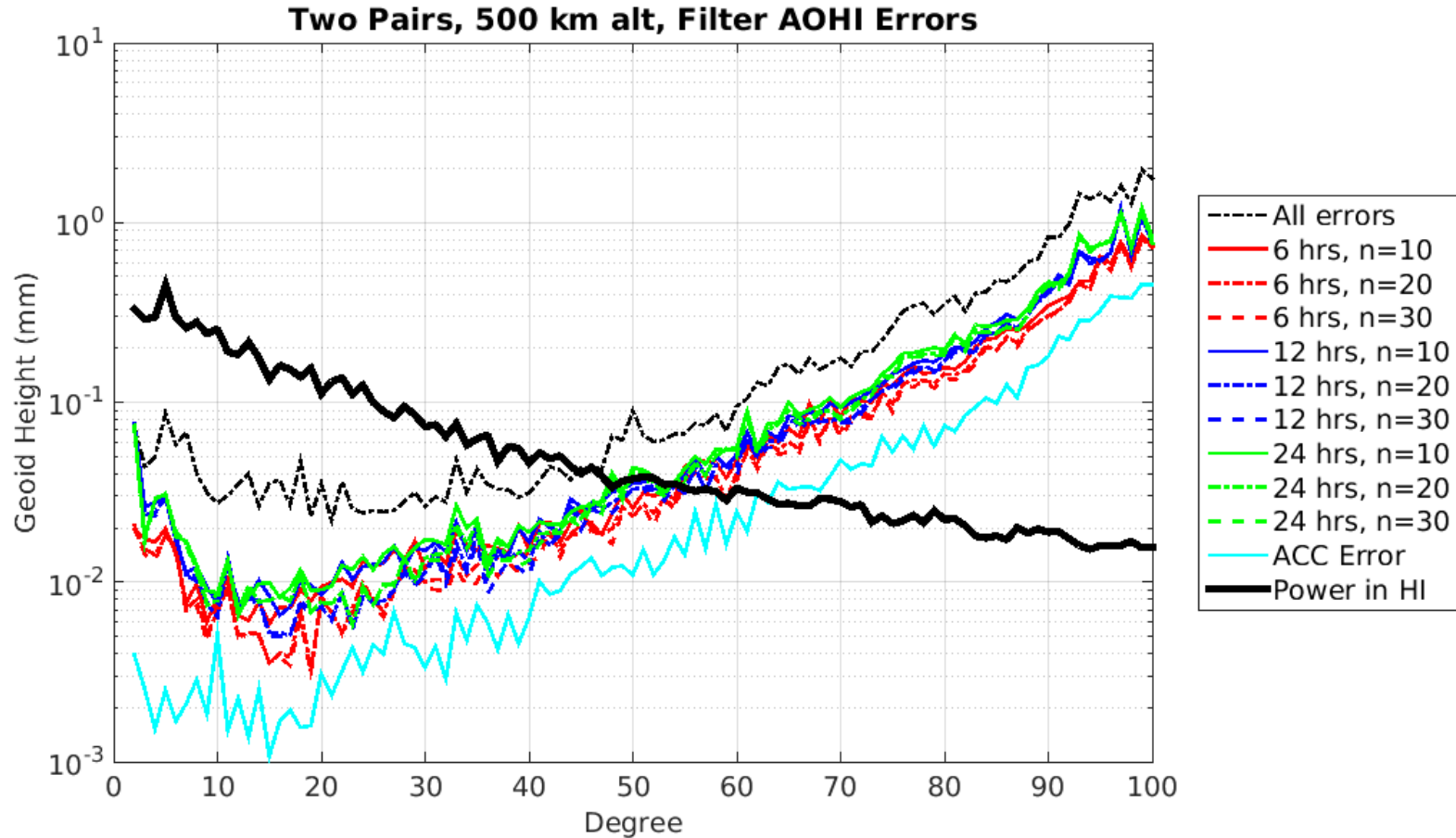
Aliasing: OT + AOHIS



Architecture: Two Pairs

Altitude: 500 km

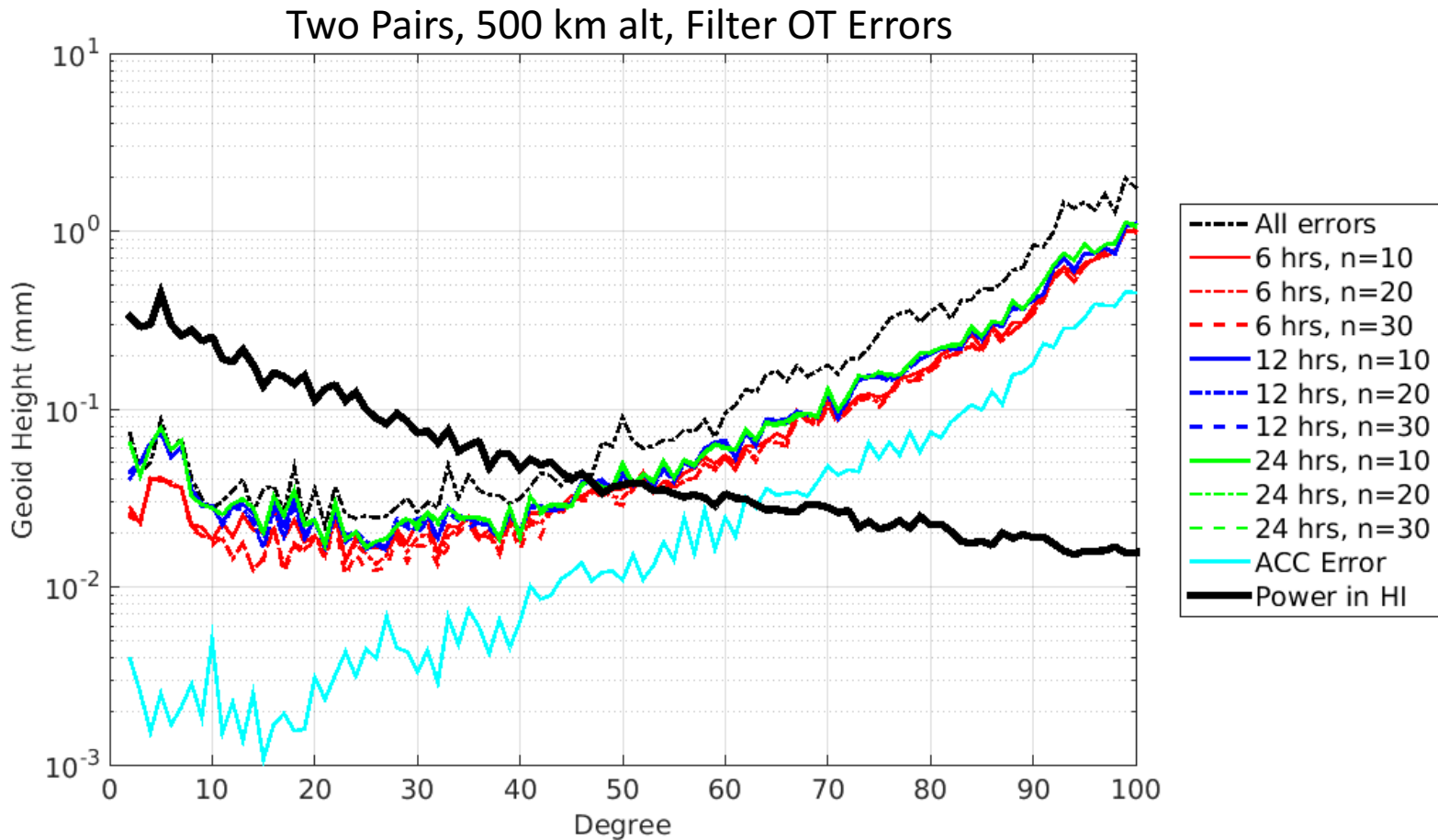
Aliasing: AOHI



Architecture: Two Pairs

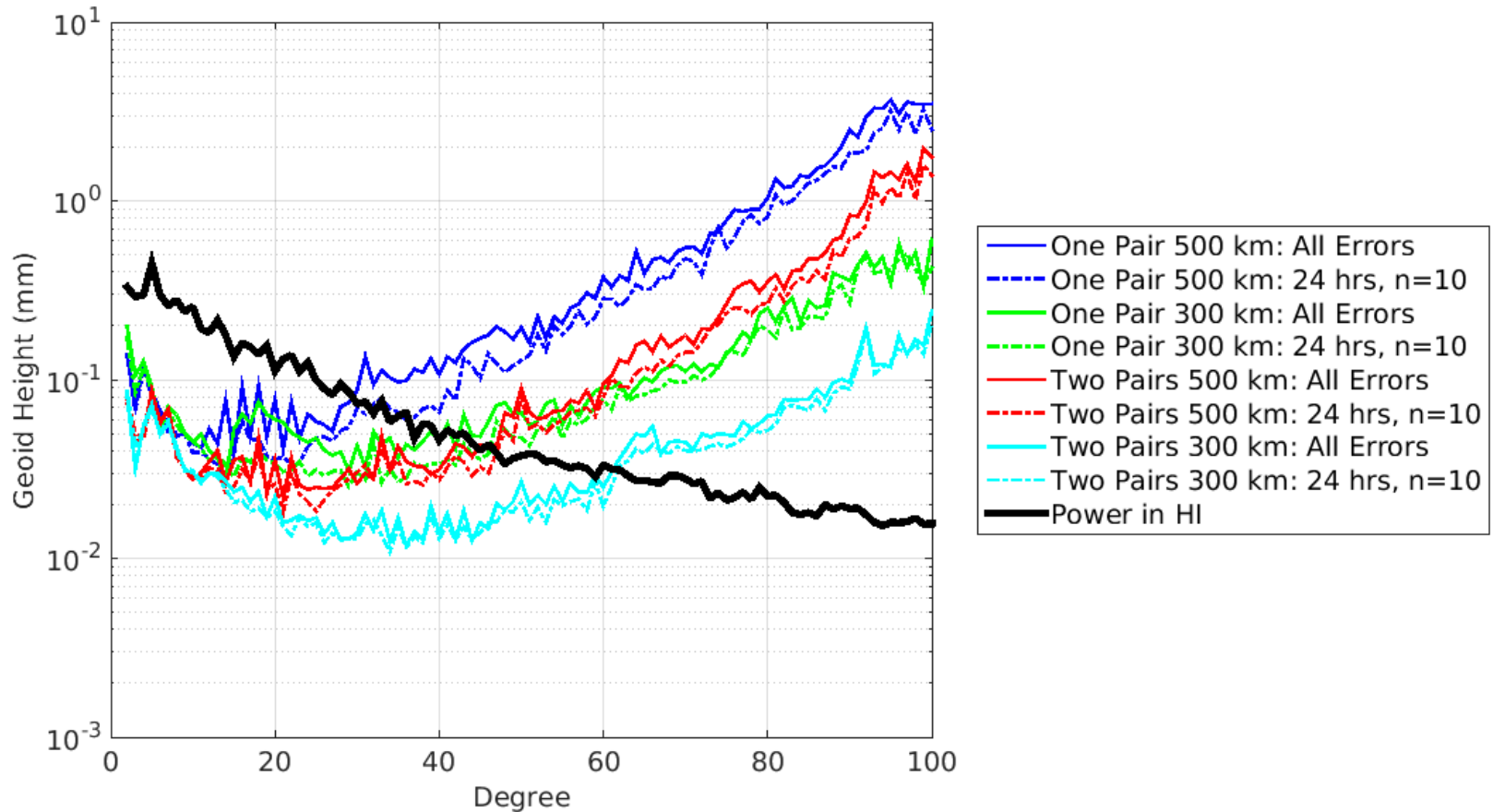
Altitude: 500 km

Aliasing: OT

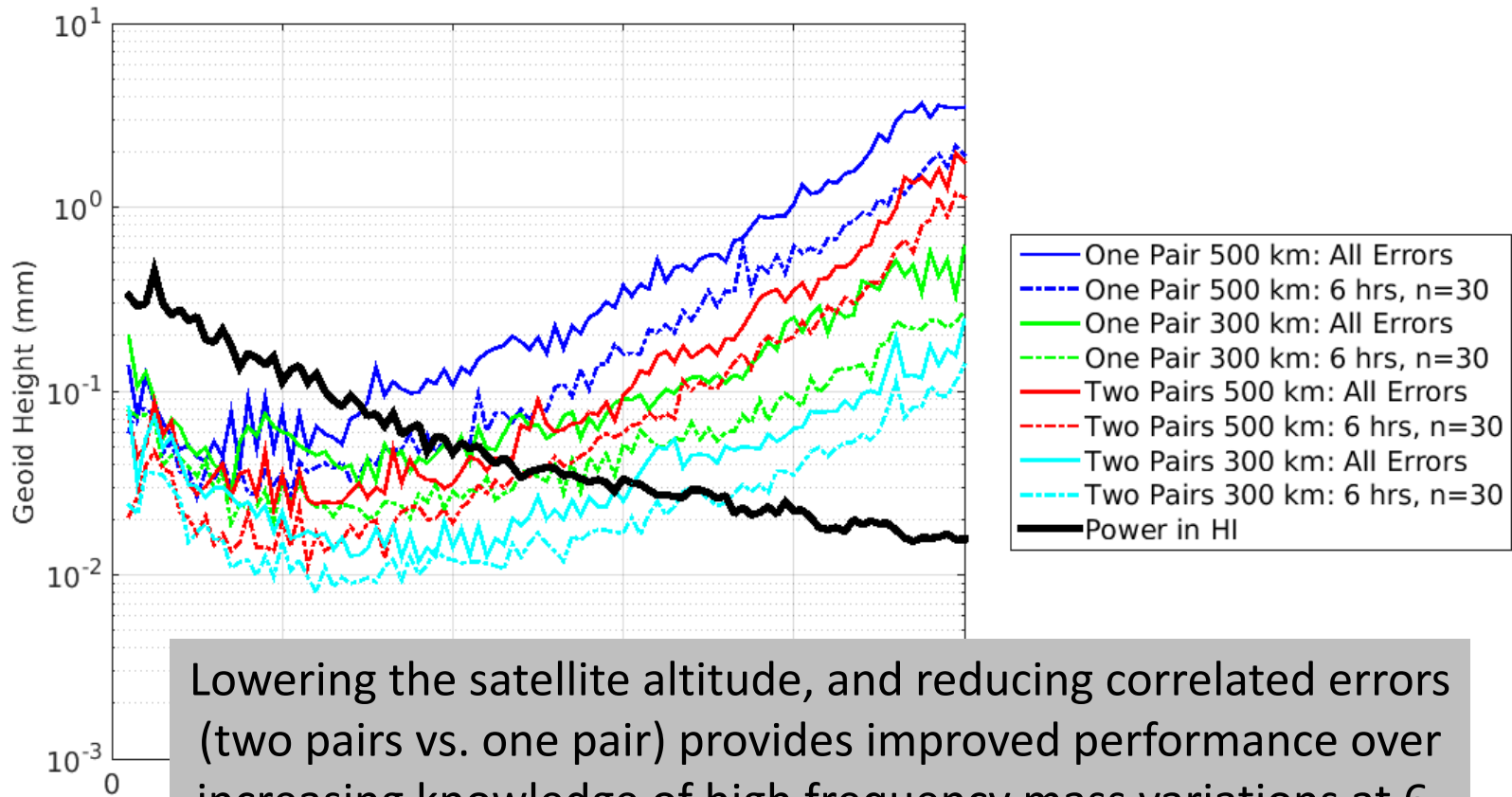




# Summary: Daily 10 x 10

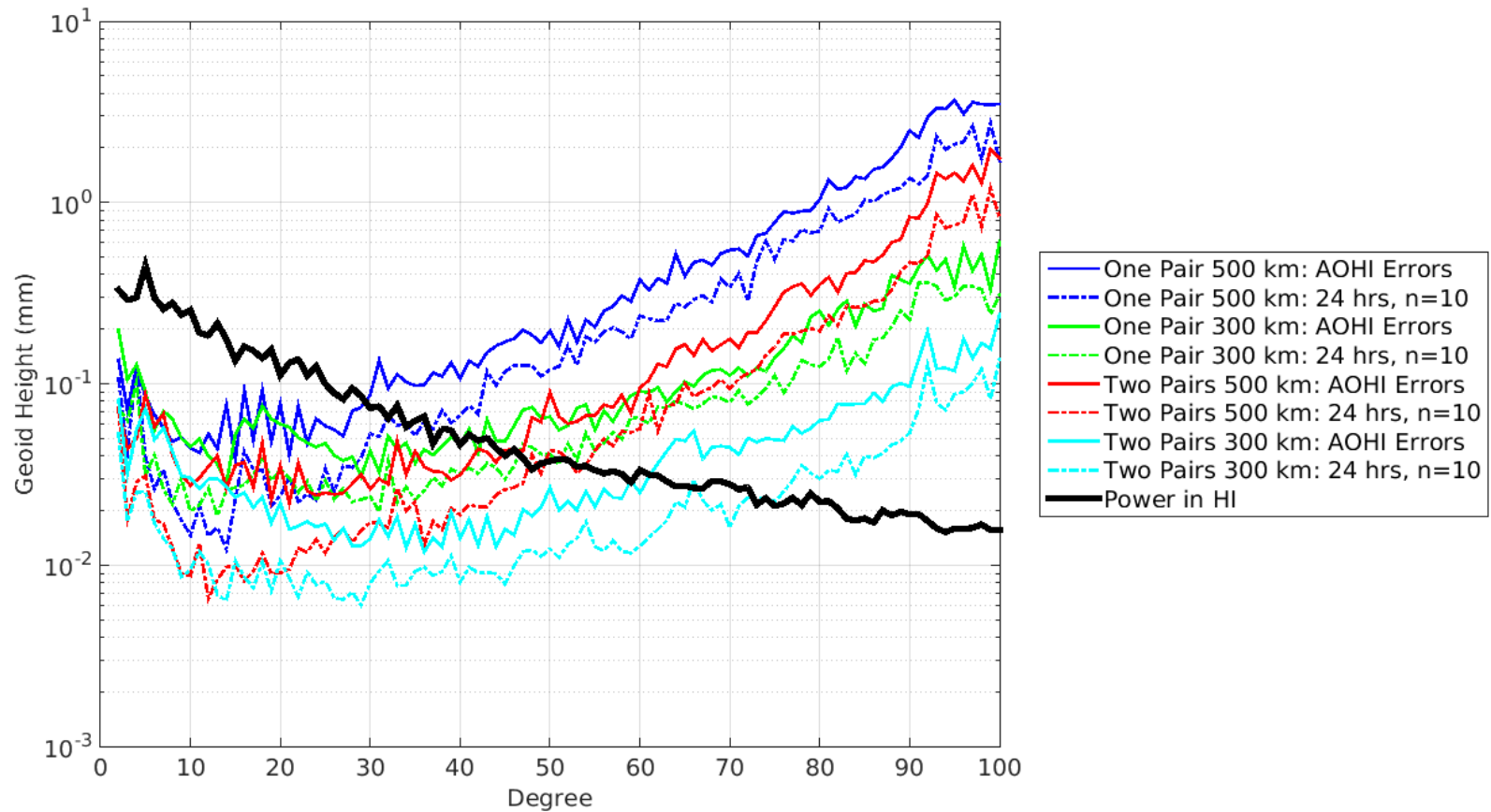


# Summary: 6-hourly 30 x 30

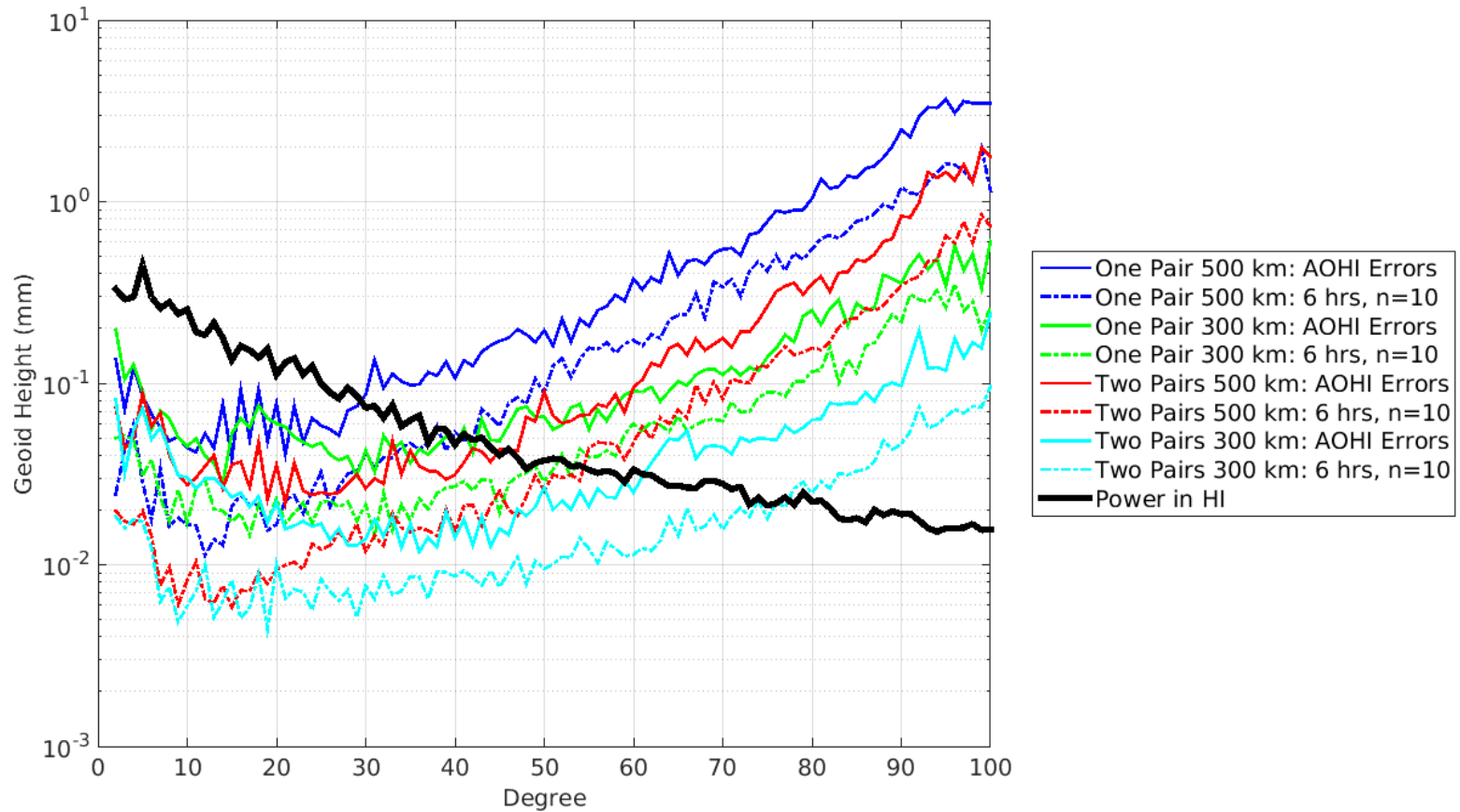


Lowering the satellite altitude, and reducing correlated errors (two pairs vs. one pair) provides improved performance over increasing knowledge of high frequency mass variations at 6-hourly, 667 km ( $n = 30$ ) spatial scales

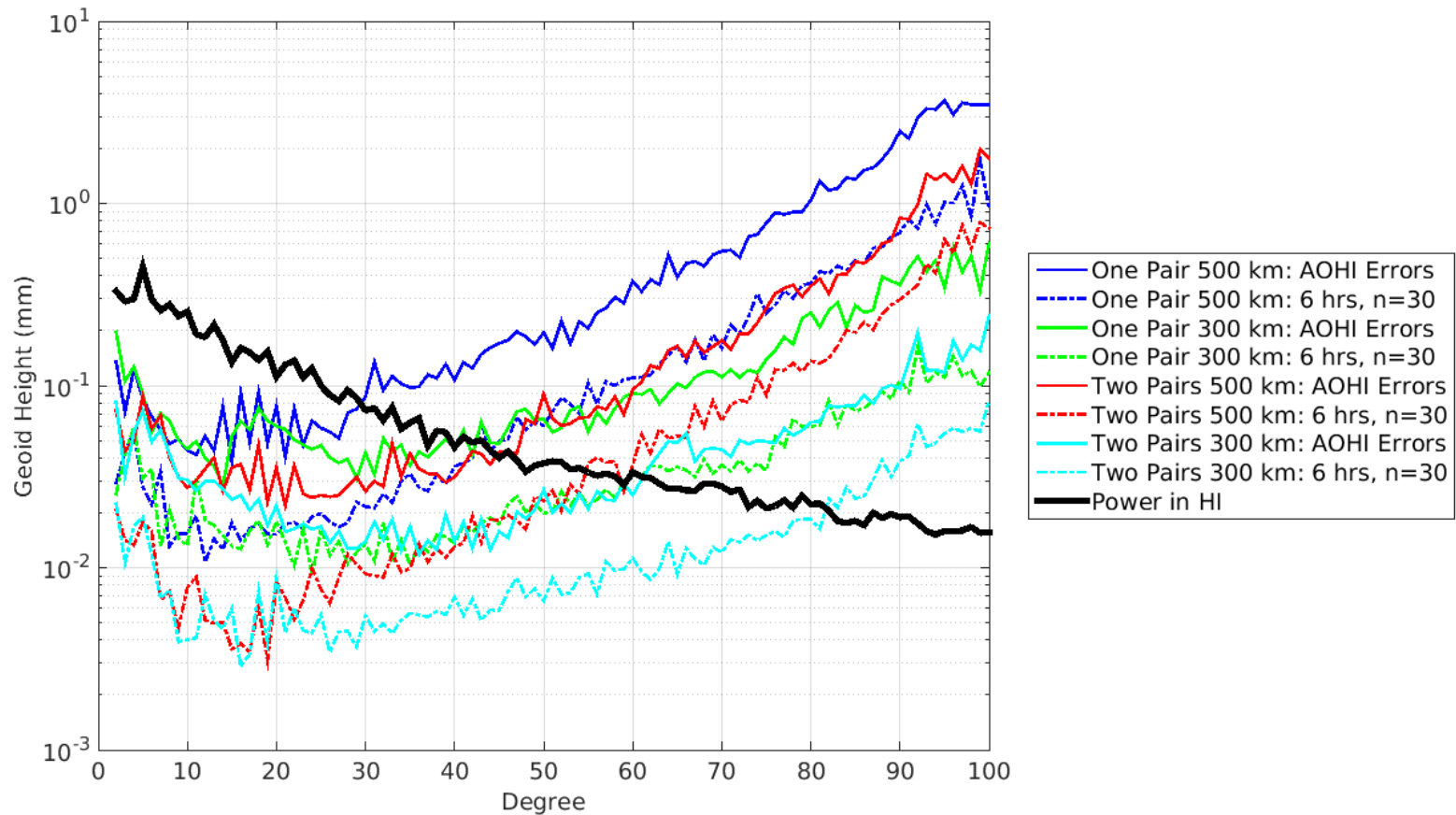
# AOHI Errors Only: Daily 10 x 10



# AOHI Errors Only: 6-hourly 10 x 10



# AOHI Errors Only: 6-hourly 30 x 30



# Summarizing Points

- Temporal aliasing errors at sub-6-hour time scales and small spatial scales ( $n > 30$ ; 667 km) are still dominant errors in the gravity recovery process
  - Large improvements in knowledge of high frequency mass variations lead to modest improvements in our ability to improve the spatial resolution of monthly gravity field estimates
- Ocean tide errors are less sensitive than AOD errors to spatial/temporal resolution of improvement in knowledge
  - This thought experiment was designed to target AOD aliasing error reduction more than ocean tide aliasing errors
  - Likely need to co-estimate ocean tides concurrently with the gravity field reduce aliasing error
- Reducing correlated error, as well as lowering the satellite altitude, has a greater impact than reducing temporal aliasing errors as examined in this study
- Speculative: Current level of performance of the accelerometers appears to be adequate for the foreseeable future, as aliasing errors will likely continue to dominate